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# Pressure observer based adaptive robust trajectory tracking control of a parallel manipulator driven by pneumatic muscles<sup>\*</sup>

ZHU Xiao-cong<sup>†</sup>, TAO Guo-liang<sup>†‡</sup>, CAO Jian

(The State Key Laboratory of Fluid Power Transmission and Control, Zhejiang University, Hangzhou 310027, China) <sup>†</sup>E-mail: zhuxiaoc@zju.edu.cn; gltao@zju.edu.cn Received Mar. 6, 2007; revision accepted Aug. 1, 2007

Abstract: This paper presents a pressure observer based adaptive robust controller (POARC) for posture trajectory tracking of a parallel manipulator driven by three pneumatic muscles without pressure sensors. Due to model errors of the static forces and friction forces of pneumatic muscles, simplified average flow rate characteristics of valves, unknown disturbances of entire system, and unmeasured pressures, there exist rather severe parametric uncertainties, nonlinear uncertainties and dynamic uncertainties in modeling of the parallel manipulator. A nonlinear pressure observer is constructed to estimate unknown pressures on the basis of a single-input-single-output (SISO) decoupling model that is simplified from the actual multiple-input-multiple-output (MIMO) coupling model of the parallel manipulator. Then, an adaptive robust controller integrated with the pressure observer is developed to accomplish high precision posture trajectory tracking of the parallel manipulator. The experimental results indicate that the system with the proposed POARC not only achieves good control accuracy and smooth movement but also maintains robustness to disturbances.

Key words:Pneumatic muscle, Parallel manipulator, Pressure observer, Adaptive robust control, Trajectory trackingdoi:10.1631/jzus.2007.A1928Document code: ACLC number: TH138

# INTRODUCTION

A pneumatic muscle is a new kind of pneumatic actuator similar to human muscle, which has been gradually applied in robotic manipulators (Caldwell *et al.*, 1995; Medrano-Cerda *et al.*, 1995; Tondu and Lopez, 2000; Nakamura *et al.*, 2002; Tsagarakis and Caldwell, 2003; Costa and Caldwell, 2006; Takuma and Hosoda, 2006; Ahn and Nguyen, 2007). The parallel manipulator driven by three pneumatic muscles studied in this paper has the advantages of cleanness, light weight, low cost, easy maintenance, compact structure and high power/volume ratio, and will have promising wide applications in robotics, industrial automation and bionic devices (Zhu and Tao, 2004; Tao *et al.*, 2005).

Due to model errors of the static forces and friction forces of pneumatic muscles, simplified average flow rate characteristics of valves, and unknown disturbances of the entire system, there exist rather severe parametric uncertainties and nonlinear uncertainties in modeling of the parallel manipulator. Recently, an adaptive robust controller has been designed to effectively deal with the above uncertainties with the guarantee of good transient performance and final tracking accuracy (Zhu et al., 2006). In order to reduce cost and complexity of the pneumatic system, pressure sensors should be used as little as possible in practice. Therefore, it is necessary to develop this controller without the need of measuring pressures for trajectory tracking of the parallel manipulator driven by pneumatic muscles. However, the absence of pressure sensors will bring a new challenge for controlling such a system and achieving good performance since pressure feedback is really needed in the previous adaptive robust controller.

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<sup>&</sup>lt;sup>‡</sup> Corresponding author

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In recent years, many researchers have worked on applying observer theory to designing controller of pneumatic systems and cancelling pressure sensors. Specially, Hildebrandt et al. (2005) proposed a feedback linearization controller based on a pressure observer for tracking control of a pneumatic system, which is realized by parameterizing the chamber pressures in respect of the actual position and corresponding derivatives. Gulati and Barth (2005) presented a sliding mode controller based on either energy based or force-error based Lyapunov pressure observer for servo control of pneumatic actuators. Pandian et al.(2002) developed a continuous gain observer and a sliding-mode observer to estimate the pressures in the cylinder, and used the estimated pressure in a sliding-mode controller for tracking control of a pneumatic cylinder. For the parallel manipulator studied, it must be noted that the system has MIMO coupling dynamics and large uncertainties associated with the pneumatic muscle, and that the unmeasured pressure could not be accurately represented by other measured state variables in linear form. Therefore, the designing method of this pressure observer is different from those in the above literature.

In this paper, informed by research on the adaptive robust observer (Yao and Xu, 2001), a nonlinear pressure observer based on a simplified SISO decoupling model with large uncertainties, is integrated with an adaptive robust controller to accomplish high precision posture trajectory tracking of the parallel manipulator without the need of measuring pressures.

# DECOUPLING DYNAMICS

The parallel manipulator driven by three pneumatic muscles (manufactured by Festo, MAS-40-N600-AA-MCKK) is shown in Fig.1, which consists of a moving platform, a base platform, a central pole and three pneumatic muscles connected by six ball joints that are evenly distributed along the respective platforms. The central pole is fixed to the base platform and is connected to the moving platform by a spherical joint. Two fast switching valves (manufactured by Festo, MHE2-MS1H-3/2G-M7-K, 100 L/min) are utilized to control the pressure inside each pneumatic muscle and this combination of components is referred to as a driving unit subsequently.



1: Moving platform; 2: Ball joint; 3: Spherical joint 4: Pneumatic muscle; 5: Central pole; 6: Base platform

Fig.1 Experimental test-rig of parallel manipulator driven by pneumatic muscles

#### SISO decoupling model in task-space

The dynamics in task-space of the parallel manipulator is given by (Zhu *et al.*, 2006; Tao *et al.*, 2007)

$$\boldsymbol{M}(\boldsymbol{\theta})\ddot{\boldsymbol{\theta}} + \boldsymbol{C}(\boldsymbol{\theta}, \dot{\boldsymbol{\theta}})\dot{\boldsymbol{\theta}} + \boldsymbol{G}(\boldsymbol{\theta}) + \boldsymbol{J}^{\mathrm{T}}(\boldsymbol{\theta})\boldsymbol{F}_{\mathrm{f}}(\boldsymbol{l}, \dot{\boldsymbol{l}}, \boldsymbol{p}) + \boldsymbol{d}_{\mathrm{t}}(\boldsymbol{t})$$
$$= \boldsymbol{J}^{\mathrm{T}}(\boldsymbol{\theta})\boldsymbol{F}_{\mathrm{m}}(\boldsymbol{p}, \boldsymbol{l}), \qquad (1)$$

where  $\theta = [\theta_x, \theta_y]^T$  is the posture vector of the parallel manipulator,  $I = [l_1, l_2, l_3]^T$  is the contractive length vector of pneumatic muscles,  $p = [p_1, p_2, p_3]^T$  is the relative pressure vector of pneumatic muscles,  $M(\theta)$  is the rotational inertial matrix,  $C(\theta, \dot{\theta})\dot{\theta}$  is the vector of centripetal and Coriolis torques,  $G(\theta)$  is the vector of gravitational torques,  $F_f(I, \dot{I}, p)$  is the friction force vector of pneumatic muscles and link-joints,  $d_t(t)$  is the disturbance vector in task-space,  $J(\theta)$  is Jacobian transformation matrix and  $F_m(p, I)$  is the static force vector of pneumatic muscles with each component given by (Tondu and Lopez, 2000)

$$F_{\rm mi}(l_i, p_i) = A(l_i)p_i + F_{\rm r}(l_i),$$
(2)

where  $A(l_i)$  and  $F_r(l_i)$  are the equivalent cylinder area and rubber elastic force of pneumatic muscle, respectively.

Substitute Eq.(2) into Eq.(1) while noting  $\dot{l} = J(\theta)\dot{\theta}$ , one obtains

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$$\boldsymbol{l} = \boldsymbol{H}_{a}(\boldsymbol{\theta})\boldsymbol{A}_{c}(\boldsymbol{l})\boldsymbol{p} + \boldsymbol{H}_{b}(\boldsymbol{\theta},\boldsymbol{\theta}) + \boldsymbol{d}_{1}, \qquad (3)$$

$$\begin{split} \boldsymbol{H}_{a}(\boldsymbol{\theta}) &= \boldsymbol{J}(\boldsymbol{\theta})\boldsymbol{M}^{-1}(\boldsymbol{\theta})\boldsymbol{J}^{\mathrm{T}}(\boldsymbol{\theta}), \\ \boldsymbol{A}_{c}(\boldsymbol{l}) &= \mathrm{diag}\big(\boldsymbol{A}(l_{1}),\boldsymbol{A}(l_{2}),\boldsymbol{A}(l_{3})\big), \\ \boldsymbol{H}_{b}(\boldsymbol{\theta}) &= \boldsymbol{J}(\boldsymbol{\theta})\boldsymbol{M}^{-1}(\boldsymbol{\theta})\boldsymbol{J}^{\mathrm{T}}(\boldsymbol{\theta})\boldsymbol{F}_{r}(\boldsymbol{l}) + \dot{\boldsymbol{J}}(\boldsymbol{\theta},\dot{\boldsymbol{\theta}})\dot{\boldsymbol{\theta}} - \\ \boldsymbol{J}(\boldsymbol{\theta})\boldsymbol{M}^{-1}(\boldsymbol{\theta})\Big[\boldsymbol{C}(\boldsymbol{\theta},\dot{\boldsymbol{\theta}})\dot{\boldsymbol{\theta}} + \boldsymbol{G}(\boldsymbol{\theta}) + \boldsymbol{J}^{\mathrm{T}}\boldsymbol{F}_{f}(\boldsymbol{l},\dot{\boldsymbol{l}},\boldsymbol{p})\Big], \\ \boldsymbol{d}_{1} &= -\boldsymbol{J}(\boldsymbol{\theta})\boldsymbol{M}^{-1}(\boldsymbol{\theta})\boldsymbol{d}_{t}. \end{split}$$

Eq.(3) is a MIMO dynamics and there exist coupling effects between the pressures and the contractive accelerations of different driving units. To remove such coupling effects, a SISO decoupling dynamics in task-space is proposed as follows for each driving unit i:

$$\ddot{l}_i = f_{a}(\boldsymbol{\theta})A(l_i)p_i + f_{b}(\boldsymbol{\theta},\dot{\boldsymbol{\theta}}) + d_{xi}, \qquad (4)$$

where  $f_a(\theta)$  is the *i*th principal diagonal element of matrix  $H_a(\theta)$ ,  $f_b(\theta, \dot{\theta})$  is the *i*th element of  $H_b(\theta, \dot{\theta})$ , and  $d_{xi}$  is the lumped disturbance including the disturbance of the *i*th driving unit and the coupling effects of the other two driving units.

# Simplified model in muscle-space

The pressure dynamics of each driving unit is (Zhu et al., 2006)

$$\dot{p}_{i} = -\frac{\lambda_{ai}(p_{i} + p_{0})\dot{V}_{i}}{V_{i}} + \frac{\lambda_{bi}RT_{i}q_{mi}}{V_{i}} + d_{mi}, \qquad (5)$$

where  $\lambda_{ai}$  and  $\lambda_{bi}$  are the polytropic exponents,  $p_0$  the atmospheric pressure,  $V_i$  the pneumatic muscle's inner volume, R the gas constant,  $T_i$  the thermodynamic temperature of pneumatic muscle,  $d_{mi}$  the disturbance in muscle-space, and  $q_{mi}$  the mass flow rate of air through the valve given by

$$q_{\rm mi} = u_i \chi_i(p_{\rm ui}, p_{\rm di}, T_{\rm ui}), \tag{6}$$

where  $u_i$  is the duty cycle and  $\chi_i$  is a nonlinear flow gain function,  $p_{ui}$  and  $p_{di}$  are the upstream pressure and downstream pressure respectively, and  $T_{ui}$  is the upstream temperature.

Considering that the pressure information of pneumatic muscle must be provided in the process of

designing the controller while the pressure could not be obtained by measurement in the absence of pressure sensors, the model-based pressure observer should be constructed according to Eqs.(5) and (6). However, since Eq.(6) is a nonlinear function of the unmeasured pressure, Eqs.(5) and (6) cannot be directly used to design the pressure observer. Hence, assuming the flow rate gain function  $\chi_l(p_{ul}, p_{dl}, T_{ul})$  to be constant and merging Eq.(6) with Eq.(5), a simplified model in muscle-space could be expressed as Eq.(7), which is a linear function of the unmeasured pressure:

$$\dot{p}_{i} = g_{a}(l_{i})u + g_{b}(l_{i})l_{i}p_{i} + g_{c}(l_{i})l_{i} + d_{pi}, \qquad (7)$$

where  $g_a(l_i)$ ,  $g_b(l_i)$  and  $g_c(l_i)$  are nonlinear functions of contractive length of pneumatic muscles:

$$g_{a}(l_{i}) = \frac{\lambda_{bi}RT_{i}\chi_{i}}{V(l_{i})}, \quad g_{b}(l_{i}) = -\frac{\lambda_{ai}}{V(l_{i})}\frac{\partial V(l_{i})}{\partial l_{i}},$$
$$g_{c}(l_{i}) = -\frac{\lambda_{ai}p_{0}}{V(l_{i})}\frac{\partial V(l_{i})}{\partial l_{i}},$$

and  $d_{pi}$  is the model error during simplification which will be attenuated by robust feedback term.

# SISO dynamics in state-space

For each driving unit, state variables are defined as  $\mathbf{x} = [l_i, \dot{l}_i, p_i]^T$  (*i*=1,2,3). According to Eqs.(4) and (7), the following SISO dynamics in state-space can be obtained:

$$\begin{aligned}
\dot{x}_{1} &= x_{2}, \\
\dot{x}_{2} &= f_{a}(\theta)A(x_{1})x_{3} + f_{b}(\theta,\dot{\theta}) + d_{x}, \\
\dot{x}_{3} &= g_{a}(x_{1})u + g_{b}(x_{1})x_{2}x_{3} + g_{c}(x_{1})x_{2} + d_{p}.
\end{aligned}$$
(8)

## PROBLEMS TO BE ADDRESSED

The nonlinear disturbances in task-space and muscle-space can be decomposed into unknown constant nominal values and time-varying uncertainties, i.e.,  $d_x = d_{x0} + \tilde{d}_x$ ,  $d_p = d_{p0} + \tilde{d}_p$ . Let  $\beta$  be the unknown parameter vector,  $\hat{\beta}$  the estimate of  $\beta$  and  $\tilde{\beta} = \hat{\beta} - \beta$  the estimation error.

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where

Assumption 1 The  $x_3$ -subsystem in Eq.(8) with  $x_3$  as the state and u,  $x_1$ ,  $x_2$  as the inputs, is bounded-inputbounded-state stable in the sense that for every  $x_{30} \in \mathbb{R}$ and every  $u, x_1, x_2 \in L^1_{\infty}[0, \infty)$ , the solution  $x_3$  starting from the initial condition  $x_{30}$  is bounded, i.e.,  $x_3(t) \in L^1_{\infty}[0, \infty)$  (Yao and Xu, 2001).

**Assumption 2** The extents of the parametric uncertainties and nonlinear uncertainties are known, i.e.,

$$\boldsymbol{\beta} \in \boldsymbol{\Omega}_{\boldsymbol{\beta}} = \left\{ \boldsymbol{\beta} : \boldsymbol{\beta}_{\min} \leq \boldsymbol{\beta} \leq \boldsymbol{\beta}_{\max} \right\}, \\ | \, \boldsymbol{\tilde{d}}_{x} \mid \leq \boldsymbol{d}_{\max}, \, | \, \boldsymbol{\tilde{d}}_{p} \mid \leq \boldsymbol{d}_{p\max},$$

where  $\boldsymbol{\beta}_{\max} = [\boldsymbol{\beta}_{1\max}, \dots, \boldsymbol{\beta}_{n\max}]^{T}$  is the maximum parameter vector and  $\boldsymbol{\beta}_{\min} = [\boldsymbol{\beta}_{1\min}, \dots, \boldsymbol{\beta}_{n\min}]^{T}$  is the minimum parameter vector,  $\boldsymbol{d}_{x\max}$  and  $\boldsymbol{d}_{p\max}$  are known vectors.

#### **Parameter projection**

A discontinuous projection can be defined as Eq.(9) in order to guarantee that the parameter estimates given by Eq.(10) remain in the known bounded region all the time (Yao and Tomizuka, 1997).

$$Proj_{\hat{\boldsymbol{\beta}}}(\delta_{i}) = \begin{cases} 0, & \hat{\boldsymbol{\beta}}_{i} = \boldsymbol{\beta}_{i\max} \text{ and } \delta_{i} > 0, \\ 0, & \hat{\boldsymbol{\beta}}_{i} = \boldsymbol{\beta}_{i\min} \text{ and } \delta_{i} < 0, \\ \delta_{i}, & \text{otherwise.} \end{cases}$$
(9)

The adaptation law is given by

$$\hat{\boldsymbol{\beta}} = \operatorname{Proj}_{\hat{\boldsymbol{\beta}}}(\boldsymbol{\Gamma}\boldsymbol{\sigma}), \tag{10}$$

where  $\Gamma > 0$  is a diagonal matrix and  $\sigma$  is an adaptation function to be synthesized later. It can be shown that for any adaptation function, the projection mapping used in Eq.(10) guarantees

(P1) 
$$\hat{\boldsymbol{\beta}} \in \Omega_{\boldsymbol{\beta}} = \{ \hat{\boldsymbol{\beta}} : \boldsymbol{\beta}_{\min} \le \hat{\boldsymbol{\beta}} \le \boldsymbol{\beta}_{\max} \},\$$
  
(P2)  $\tilde{\boldsymbol{\beta}}^{\mathrm{T}}[\boldsymbol{\Gamma}^{-1}Proj_{\hat{\boldsymbol{\beta}}}(\boldsymbol{\Gamma}\boldsymbol{\sigma}) - \boldsymbol{\sigma}] \le \mathbf{0}, \ \forall \boldsymbol{\sigma}.$  (11)

#### Difficulties in designing observer/controller

For trajectory tracking control of the parallel manipulator driven by pneumatic muscles without the need of measuring pressures, the main difficulties are analyzed as follows.

(1) Due to the nonlinearities of robotic dynamics, pressure dynamics and flow rate characteristics, the

unmeasured pressure could not be accurately represented by other measured state variables in linear form. Thus, it is impossible to design a linear pressure observer based on the accurate model (Brogan, 1985).

(2) It must be noted that there exist large uncertainties in the system dynamics, such as the parametric uncertainties from modeling the static force and the friction force of pneumatic muscle, the nonlinear uncertainties from simplifying pressure dynamics of pneumatic muscle and flow characteristic of fast switching valves and the coupling effects of MIMO dynamics, and the dynamic uncertainties from estimating the unmeasured pressures. The above uncertainties could not be attenuated only by robust observer-controller method (Tarek and Francoise, 1999). Therefore, an adaptive robust observer-controller method should be adopted for compensating and attenuating these uncertainties.

(3) The model uncertainties are mismatched, i.e., both parametric uncertainties and nonlinear uncertainties appear in the dynamic equations that are not directly related to the control input *u*. Therefore the backstepping design technology should be employed to overcome the design difficulties for achieving asymptotic stability (Bu and Yao, 2001).

#### PRESSURE ESTIMATATION

Define a variable as follows (Yao and Xu, 2001):

$$\zeta = x_3 - \omega(x_1, x_2), \tag{12}$$

where  $\omega(x_1,x_2)$  is a design function yet to be determined. From Eq.(8), the derivative of Eq.(12) is

$$\dot{\zeta} = g_{a}(x_{1})u + \left[g_{b}(x_{1})x_{2} - \frac{\partial\omega}{\partial x_{2}}f_{a}(\theta)A(x_{1})\right](\zeta + \omega) + g_{c}(x_{1})x_{2} + d_{p} - \frac{\partial\omega}{\partial x_{1}}x_{2} - \frac{\partial\omega}{\partial x_{2}}f_{b}(\theta, \dot{\theta}) - \frac{\partial\omega}{\partial x_{2}}d_{x}.$$
 (13)

For simplicity, let

$$\begin{cases} A_{\zeta} = g_{b}(x_{1})x_{2} - \frac{\partial\omega}{\partial x_{2}}f_{a}(\theta)A(x_{1}), \zeta_{\eta 01} = g_{a}(x_{1})u, \\ \zeta_{\eta 02} = A_{\zeta}\omega + g_{c}(x_{1})x_{2} - \frac{\partial\omega}{\partial x_{1}}x_{2} - \frac{\partial\omega}{\partial x_{2}}f_{b}(\theta,\dot{\theta}), \quad (14) \\ \zeta_{\eta 1} = -\frac{\partial\omega}{\partial x_{2}}, \quad \zeta_{\eta 2} = 1. \end{cases}$$

Then, Eq.(13) is rewritten as

$$\dot{\zeta} = A_{\zeta}\zeta + \zeta_{\eta 01} + \zeta_{\eta 02} + \zeta_{\eta 1}d_{x0} + \zeta_{\eta 2}d_{p0} + \zeta_{\eta 1}\tilde{d}_{x} + \zeta_{\eta 2}\tilde{d}_{p}.$$
(15)

If  $d_{x0}$  and  $d_{p0}$  were known, a nonlinear observer would be designed as

$$\dot{\zeta} = A_{\zeta} \dot{\zeta} + \zeta_{\eta 01} + \zeta_{\eta 02} + \zeta_{\eta 1} d_{x0} + \zeta_{\eta 2} d_{p0}.$$
 (16)

Then, the state estimation error  $\tilde{\zeta} = \hat{\zeta} - \zeta$  would be governed by the following dynamic system:

$$\tilde{\zeta} = A_{\zeta}\tilde{\zeta} - \Delta_{\rm I}, \qquad (17)$$

where

$$\Delta_{\mathrm{l}} = \zeta_{\eta \mathrm{l}} \tilde{d}_{\mathrm{x}} + \zeta_{\eta \mathrm{2}} \tilde{d}_{\mathrm{p}}.$$

Since  $d_{x0}$  and  $d_{p0}$  are unknown, the observer Eq.(16) is not implementable, but it provides motivation for the design of the following nonlinear filters:

$$\begin{cases} \dot{\zeta}_{\theta 01} = A_{\zeta} \zeta_{\theta 01} + \zeta_{\eta 01}, \quad \dot{\zeta}_{\theta 02} = A_{\zeta} \zeta_{\theta 02} + \zeta_{\eta 02}, \\ \dot{\zeta}_{\theta 1} = A_{\zeta} \zeta_{\theta 1} + \zeta_{\eta 1}, \quad \dot{\zeta}_{\theta 2} = A_{\zeta} \zeta_{\theta 2} + \zeta_{\eta 2}. \end{cases}$$
(18)

The pressure estimation can thus be represented by

$$\hat{\zeta} = \zeta_{\theta 01} + \zeta_{\theta 02} + \zeta_{\theta 1} d_{x0} + \zeta_{\theta 2} d_{p0}.$$
 (19)

From Eqs.(18) and (19), it can be verified that the observer error dynamics is still described by Eq.(17). Therefore, the unmeasured pressure is

$$x_3 = \zeta_{\theta 01} + \zeta_{\theta 02} + \zeta_{\theta 1} d_{x0} + \zeta_{\theta 2} d_{p0} + \omega - \tilde{\zeta}.$$
 (20)

Let  $A_{\zeta} = -k$  (k > 0) such that the unperturbed system of observer error dynamics is exponentially stable, i.e., when  $\Delta_{l} = 0$ , the observation error  $\zeta \tilde{\zeta}$ converges to zero exponentially. Thus, from Eq.(14),  $\omega(x_1, x_2)$  is obtained.

$$\omega(x_1, x_2) = n_{\rm a}^{-1}(\theta, x_1) [g_{\rm b}(x_1) x_2^2 / 2 + k x_2], \quad (21)$$

where  $n_a(\theta, x_1) = f_a(\theta) A(x_1)$  is a positive function.

Hence, according to Assumption 1, the observation error would be bounded.

# ADAPTIVE ROBUST CONTROLLER DESIGN

A discontinuous projection based adaptive robust controller integrated with the above pressure observer is developed to accomplish the high precision posture trajectory tracking of the parallel manipulator with dynamic uncertainties (Yao and Tomizuka, 1997; Yao and Xu, 2001). The procedure of backstepping design is illustrated as follows.

Step 1: Define a switching-function-like quantity as

$$z_2 = \dot{z}_1 + k_c z_1, \tag{22}$$

where  $z_1=x_1-x_{1d}$  is the trajectory tracking error and  $k_c$  is a positive feedback constant. If  $z_2$  converges to a small value or zero, then  $z_1$  will converge to a small value or zero since the transfer function from  $z_2$  to  $z_1$  is stable. Substituting Eqs.(8) and (20) into Eq.(22),  $z_2$  dynamics is

$$\dot{z}_{2} = n_{\mathrm{a}}\zeta_{\theta 01} + n_{\mathrm{a}}\zeta_{\theta 02} + (n_{\mathrm{a}}\zeta_{\theta 1} + 1)d_{\mathrm{x0}} + n_{\mathrm{a}}\zeta_{\theta 2}d_{\mathrm{p0}} + n_{\mathrm{a}}\omega - n_{\mathrm{a}}\tilde{\zeta} + f_{\mathrm{b}} - \ddot{x}_{\mathrm{d}} + k_{\mathrm{c}}\dot{e}.$$
(23)

For the purpose of  $z_2$  converging to zero, define the unknown parameters in task-space as  $\boldsymbol{\beta}_2 = [d_{x0}, d_{p0}]^T$ and the virtual input as  $v = \zeta_{\theta 01}$ . Then, the regressor for parameter adaptation is  $\boldsymbol{\varphi}_2 = [n_a \zeta_{\theta 1} + 1, n_a \zeta_{\theta 2}]^T$ . And  $\hat{\boldsymbol{\beta}}_2$ is updated by  $\dot{\boldsymbol{\beta}}_2 = Proj_{\hat{\boldsymbol{\beta}}}(\boldsymbol{\Gamma}_2 \boldsymbol{\sigma}_2)$  with the parameter adaptation function given by  $\boldsymbol{\sigma}_2 = \boldsymbol{\varphi}_2 z_2$ .

The desired virtual input consists of two terms:

$$v_{\rm d} = v_{\rm da} + v_{\rm ds}, \qquad (24a)$$

$$v_{\rm da} = n_{\rm a}^{-1} [-n_{\rm a} \zeta_{\theta 02} - n_{\rm a} \omega - f_{\rm b} + \ddot{x}_{\rm d} - k_{\rm c} \dot{e} - \boldsymbol{\varphi}_2^{\rm T} \hat{\boldsymbol{\beta}}_2], (24b)$$

where  $v_{da}$  functions as the adaptive control part used to achieve an improved model compensation, and  $v_{ds}$ is a robust control law including the following two terms:

$$v_{\rm ds} = v_{\rm ds1} + v_{\rm ds2}, \quad v_{\rm ds1} = -n_{\rm a}^{-1}k_2 z_2,$$
 (25)

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where  $k_2$  is a positive definite control gain,  $v_{ds2}$  is synthesized to dominate the model uncertainties coming from both parametric uncertainties and nonlinear uncertainties, which is chosen to satisfy the following conditions:

$$\begin{cases} z_2(n_a v_{ds2} - \boldsymbol{\varphi}_2^{\mathsf{T}} \tilde{\boldsymbol{\beta}}_2 - n_a \tilde{\boldsymbol{\zeta}}) \leq \varepsilon_2, \\ z_2 v_{ds2} \leq 0, \end{cases}$$
(26)

where  $\varepsilon_2$  is a positive design parameter.

Define the positive semi-definite (p.s.d) function  $V_2 = z_2^2/2$  and denote the input discrepancy as  $z_3 = v - v_d$ . From Eqs.(23) and (24), the time derivative of  $V_2$  is

$$\dot{V}_{2} = -k_{2}z_{2}^{2} + z_{2}[-\boldsymbol{\varphi}_{2}^{T}\tilde{\boldsymbol{\beta}}_{2} + n_{a}z_{3} - n_{a}\tilde{\boldsymbol{\zeta}} + n_{a}v_{ds2}].$$
(27)

Step 2: Synthesize a control input u so that  $z_3$  converges to zero or a small value with a guaranteed transient performance.

The time derivative of  $z_3$  is given by Eq.(28) while noting Eqs.(14), (18) and (24):

$$\dot{z}_{3} = A_{\zeta} \zeta_{\theta 01} + g_{a} u - \dot{v}_{dc} - \dot{v}_{du}, \qquad (28)$$

where

$$\dot{v}_{dc} = \frac{\partial v_{d}}{\partial x_{1}} \hat{x}_{2} + \frac{\partial v_{d}}{\partial x_{2}} \hat{x}_{2} + \frac{\partial v_{d}}{\partial \omega} \left( \frac{\partial \omega}{\partial x_{1}} \hat{x}_{2} + \frac{\partial \omega}{\partial x_{2}} \hat{x}_{2} \right) + \frac{\partial v_{d}}{\partial \zeta_{\theta 0 2}} \dot{\zeta}_{\theta 0 2} + \frac{\partial v_{d}}{\partial \zeta_{\theta 1}} \dot{\zeta}_{\theta 1} + \frac{\partial v_{d}}{\partial \zeta_{\theta 2}} \dot{\zeta}_{\theta 2} + \frac{\partial v_{d}}{\partial \hat{\beta}_{2}} \dot{\hat{\beta}}_{2} + \frac{\partial v_{d}}{\partial t},$$

and

$$\dot{v}_{du} = \left(\frac{\partial v_d}{\partial x_1} + \frac{\partial v_d}{\partial \omega}\frac{\partial \omega}{\partial x_1}\right)(x_2 - \hat{x}_2) + \left(\frac{\partial v_d}{\partial x_2} + \frac{\partial v_d}{\partial \omega}\frac{\partial \omega}{\partial x_2}\right)(\dot{x}_2 - \dot{x}_2),$$

where  $\hat{x}_2$  and  $\dot{x}_2$  are deduced from  $x_1$  by a second-order differential filter (Zhu *et al.*, 2006). Note that  $\dot{v}_{dc}$  represents the calculable part of  $\dot{v}_d$  and can be used to design control functions, but  $\dot{v}_{du}$  can not due to various uncertainties.

To attenuate the effect of  $\dot{v}_{du}$ , let  $\dot{v}_{du}$  be decomposed into a constant nominal value and time-varying uncertainties, i.e.,  $\dot{v}_{du} = d_{v0} + \tilde{d}_{v}$ . For the purpose of  $z_3$  converging to zero or a small value with a guaranteed performance, define the unknown parameter in muscle-space as  $\beta_3 = d_{v0}$ . Then, the regressor for parameter adaptation is  $\varphi_3 = 1$ . And  $\hat{\beta}_3$  is updated by  $\dot{\beta}_3 = Proj_{\hat{\beta}}(\Gamma_3\sigma_3)$  with the parameter adaptation function given by  $\sigma_3 = -\varphi_3 z_3$ .

The desired control input consists of two terms:

$$u = u_{da} + u_{ds}, \ u_{da} = g_a^{-1} (-A_{\zeta} \zeta_{\theta 01} + \dot{v}_{dc} - n_a z_3 + \hat{d}_{v0}), (29)$$

where  $u_{da}$  is used for adaptive model compensation and the robust control law  $u_{ds}$  consists of the following two terms:

$$u_{\rm ds} = u_{\rm ds1} + u_{\rm ds2}, \ u_{\rm ds1} = -g_{\rm a}^{-1}k_3 z_3,$$
 (30)

where  $k_3$  is a positive feedback gain,  $u_{ds2}$  is a robust control function chosen to satisfy the following conditions to dominate all model uncertainties:

$$\begin{cases} z_3(g_a u_{ds2} + \boldsymbol{\varphi}_3^T \tilde{\boldsymbol{\beta}}_3 - \tilde{d}_v) \le \varepsilon_3, \\ z_3 u_{ds2} \le 0, \end{cases}$$
(31)

where  $\varepsilon_3$  is a positive design parameter which can be arbitrarily small.

To see how the above control function works, define a p.s.d. function  $V_3 = V_2 + z_3^2/2$ . The time derivative of  $V_3$ , when Eqs.(28) and (29) are substituted into, is

$$\dot{V}_{3} = -k_{2}z_{2}^{2} - k_{3}z_{3}^{2} + z_{3}[g_{a}u_{ds2} + \boldsymbol{\varphi}_{3}^{T}\tilde{\boldsymbol{\beta}}_{3} - \tilde{d}_{v}] + z_{2}[-\boldsymbol{\varphi}_{2}^{T}\tilde{\boldsymbol{\beta}}_{2} - n_{a}\tilde{\boldsymbol{\zeta}} + n_{a}v_{ds2}].$$
(32)

Substituting Eqs.(26) and (31) into Eq.(32), then  $V_3$  is bounded above by

$$\dot{V}_3 \leq -k_2 z_2^2 - k_3 z_3^2 + \varepsilon_2 + \varepsilon_3.$$
 (33)

The solution of inequality (33) satisfies

$$V_3(t) \le \exp(-\lambda_v t) V_3(0) + \frac{\varepsilon_v}{\lambda_v} [1 - \exp(-\lambda_v t)], \quad (34)$$

where  $\lambda_v = 2 \times \min\{k_2, k_3\}, \varepsilon_v = \varepsilon_2 + \varepsilon_3$ .

The parameters  $k_2$ ,  $k_3$ ,  $\varepsilon_2$ ,  $\varepsilon_3$  could be designed to guarantee that the tracking error be bounded in a preset ball all the time, whereas, in fact, these pa-

rameters are constrained by the bandwidth of the total control system due to unmodeled high frequency dynamics, saturation of control inputs, and sampling frequency of digital implementation, etc.

If after a finite time,  $\tilde{d}_x = \tilde{d}_p = \tilde{d}_v = 0$ , i.e., in the presence of parametric uncertainties and dynamic uncertainties only, then asymptotic output tracking (or zero final tracking error) is also achieved.

## **RESULTS AND DISCUSSION**

The pressure observer based adaptive robust controller (POARC) is implemented on the parallel manipulator driven by pneumatic muscles without pressure sensors. Experimental results are illustrated as follows.

The controller is first tested for a smooth step response with initialized and generated trajectory shown in Fig.2. The estimates of lumped disturbances and pressures are shown in Fig.3. The steady-state errors are  $e_{xF}=0.03852^{\circ}$  and  $e_{yF}=0.03317^{\circ}$  and the maximal absolute values of the tracking errors are  $e_{xM}=0.93118^{\circ}$  and  $e_{yM}=0.73650^{\circ}$ . Despite the fact that there exist rather severe uncertainties in task-space and muscle-space due to model simplification and nonlinearties associated with the pneumatic muscle system, the satisfactory transient performance and excellent steady-state errors could be achieved due



Fig.2 Smooth step response of POARC

to the lumped disturbances and pressures being estimated and the unknown uncertainties being attenuated.

For tracking a sinusoidal posture trajectory (amplitude  $\theta_x=2^\circ$ ,  $\theta_y=5^\circ$  and period 15 s), response comparison between the POARC without pressure sensors and the adaptive robust controller (ARC) with pressure sensors (Zhu et al., 2006) are shown in Fig.4. The control inputs of POARC are shown in Fig.5, and the estimated pressures and measured pressures are shown in Fig.6. As can be seen from Fig.4, the average tracking errors of POARC are  $L_2[e_x]=0.07146^\circ$  and  $L_2[e_y]=0.15243^\circ$ , and the maximal absolute values of the tracking errors are  $e_{xM}=0.20250^{\circ}$  and  $e_{yM}=$ 0.44423°. It is obvious that the maximal absolute values of the tracking errors of POARC are a little larger than those of ARC. In Figs.5 and 6, the control inputs and the estimates of pressures are bounded all the time and the movement is smooth without control chattering since the discontinuous projection based adaptive robust controller is adopted. Though there are fairly large errors between estimated pressures and measured pressures, the tracking errors could be always small due to large parametric uncertainties and large dynamic uncertainties being compensated through using POARC.

Fig.7 shows the error responses of the contractive lengths of pneumatic muscles both with POARC and with sliding mode controller (SMC) (Tao *et al.*, 2005) under the condition of tracking the same trajectory



Fig.3 Estimates of disturbance and pressures under smooth step response.  $d_{xi}$ ,  $p_i$  and  $u_i$  (*i*=1,2,3) represent the disturbance, pressure and control input in the *i*th driving unit, respectively



Fig.4 Sinusoidal tracking response of POARC and ARC



Fig.6 Control inputs of POARC under sinusoidal trajectory

as above. It must be noted that neither of the two controllers uses pressure sensors. Obviously, the errors of the contractive lengths with SMC in Fig.7 have severe vibration and the parallel manipulator is suffering from vibration all the time since SMC utilizes the tracking errors to design the controller regardless of system model and pressure estimations. Consequently, the fast switching valves switch continuously, which will result in noises of the parallel manipulator and reduce useful life of the fast switching valves. In contrast, POARC makes full use of the available structural information of the unmeasured state dynamics and the prior knowledge about the parameter bounds to design the controller. As a result, small



Fig.5 Estimated and measured pressures with POARC under sinusoidal trajectory



Fig.7 Error comparison of contractive length with (a) POARC and (b) SMC

tracking errors with smooth movement and little control chattering are achieved.

For testing the robustness of POARC, the position transducers are given a sudden dither at t=12 s, which can be regarded as a sudden large output disturbance to the system. As can be seen from Fig.8, the system experiences large tracking errors due to the wrong feedback information of position transducers when the dither is introduced. But after the dither disappeared, the system comes back to the stable posture quickly with no fluctuation. This demonstrates the robustness of the proposed control algorithm to disturbances.



Fig.8 Robustness of POARC to sudden disturbance

#### CONCLUSION

A nonlinear pressure observer based adaptive robust controller is developed for trajectory tracking control of a parallel manipulator driven by pneumatic muscles without pressure sensors. The nonlinear pressure observer is constructed to recover the unknown states, i.e., pressures on the basis of a SISO decoupling model which is simplified from the actual MIMO coupling model. A robust filter structure is utilized to provide the practical pressure estimation. By integrating the adaptive robust control with the pressure observer, the parametric uncertainties, nonlinear uncertainties coming from model simplification and dynamics uncertainties coming from pressure estimation errors, are effectively compensated and attenuated.

The pressure observer based adaptive robust controller (POARC) is proved to be effective by experimental results. The steady-state errors are less than 0.04° under a smooth step response and the average tracking errors less than 0.16° under a sinusoidal trajectory. Compared with SMC, the proposed POARC performs much better with smooth movement and without control chattering. At the same time, it must be noted that the tracking errors remain small and are not influenced by large estimation errors of the observer.

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